

## Modular Connector for Resilient Grid Shell Structures

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### ABSTRACT

In the design, fabrication, and erection of grid shells, a major challenge is the nodal connector between members. Designers typically select a structure’s form and then design the nodal connectors to meet geometric and structural demands. However, this can lead to complicated connections that are difficult and expensive to fabricate. Each connector is also often unique. To address these challenges, this paper investigates a new approach in which a novel modular connector is designed for ease of fabrication and erection, and then structural forms are developed that use the connector repeatedly to join wide flange steel members via splice connections in double shear. This approach “modularizes” the nodal connector, which is a prefabricated, steel connector with starter segments that include webs and flanges. The flanges and webs of the modular connector and the members are joined independently thus achieving a moment-resisting connection. This facilitates truss-like, membrane-like, or beam-like behavior, as well as allows load to be redistributed in the case of sudden member loss or replacement thereby providing enhanced resiliency. Variability in form is achieved by bending the flange splice plates. This paper investigates the modular connector for free-form undulating grid shells and for rational structural forms developed through a proposed form-finding methodology. The proposed methodology relies on thrust network analysis coupled with geometric and structural constraints. The promise of the modular connector and the proposed methodology is demonstrated through finite element numerical analyses.

**Key Words:** Modular construction, Connection design, Grid shells, Form-finding, Steel structures

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## 22 **Introduction**

23 Shell structures (e.g., space frames, grid shell, domes) are appealing to architects and engineers  
24 due to their efficiency in spanning long distances while providing aesthetically pleasing forms.  
25 These systems can be continuous surfaces (e.g., thin shell concrete structures) or, as in the case  
26 of grid shells, be comprised of discrete elements (members and nodal connectors) following the  
27 geometry of a continuous surface. The geometry of grid shells can be categorized as free-form  
28 or form-found. Free-form grid shells are typically developed through computer aided software  
29 primarily following a geometrical formulation, as opposed to prioritizing a structural behavior.  
30 However, with the advancement in computational capabilities and manufacturing technologies, the  
31 ability to generate and analyze a wide range of shapes has enabled a growth in popularity of these  
32 free-form structures. Knippers and Helbig (2009) presented two examples of free-form glazed  
33 structures including the design process as well as fabrication and construction strategies. Schlaich  
34 and Schober (2005) described the development of free-form glazed structures using triangular and  
35 quadrilateral glass panels based on their long experience in design. Glymph et al. (2004) proposed  
36 a parametric strategy for the design of the Jerusalem Museum of Tolerance which is comprised of  
37 multiple free-form glass structures. Using geometric principles incorporated into computational  
38 models, Glymph et al. (2004) were able to achieve the structural form using quadrilateral glass  
39 panels. Alternatively, the shape of rational (i.e., based on structural performance) grid shells can  
40 be developed through numerical form-finding methods. Form-finding is a process that solves a  
41 system of equilibrium equations to find an optimized structural shape based on input parameters  
42 including: (1) initial geometry (or mesh), (2) boundary conditions, and (3) external loading. These  
43 input parameters can also be controlled to allow for different grid shells to be explored (Adri-  
44 aenssens et al., 2014). Numerical form-finding methods such as Force Density Method (FDM)  
45 (Linkwitz and Schek, 1971; Schek, 1974), Dynamic Relaxation (DR) (Day, 1965), and Thrust  
46 Network Analysis (TNA) (Block, 2009) have been successfully implemented in the development  
47 of rational structural forms. FDM has been mostly used for finding the form of pre-stressed cable-  
48 net roofs, hanging structures, and timber grid shells. Over the years, various modifications to the

49 method have been introduced. For example, Ye et al. (2012) proposed a modified FDM for form-  
50 finding of membrane structures that, instead of prescribing the force densities, uses the membrane  
51 stresses and cables tension force as initial conditions. Sanchez et al. (2007) introduced a multi-  
52 step FDM for form-finding tensile membrane structures by adjusting the force densities to achieve  
53 a smooth or uniform stress distribution in the final equilibrated shape. Miki and Kawaguchi (2010)  
54 proposed an extension to the FDM based on the variational principle to address limitation of the  
55 original FDM when the structure is subjected to both tension and compression. The DR method  
56 was first introduced by Day (1965) as a new approach for solving structural problems such as portal  
57 frames, flat plates, and thick cylinders. DR was later adopted for form-finding of tension structures  
58 and grid shells (Barnes, 1988, 1999). Adriaenssens and Barnes (2001) developed a method based  
59 on DR for form-finding tensegrity structures with curved splines that incorporates the effect of the  
60 bending moments. Richardson et al. (2013) proposed a two-phase approach for the design of sin-  
61 gle layer grid shells that consists of developing the initial grid shell form through DR and using a  
62 genetic algorithm to optimize the grid topology. Bagrianski and Halpern (2014) proposed a new ap-  
63 proach - Prescriptive Dynamic Relaxation - for form-finding compressive structures with initially  
64 prescribed member lengths. The TNA method was developed mostly for the design and analysis of  
65 masonry or concrete vaulted structures and is used for finding compression only structures (Block  
66 and Ochsendorf, 2007; Block, 2009). TNA has been modified and improved to include different  
67 constraints and make the form-finding algorithm faster and more robust (Fraternali, 2010; Panozzo  
68 et al., 2013; Marmo and Rosati, 2017; Liew et al., 2018). A thorough description of these methods  
69 can be found in Adriaenssens et al. (2014) and a comparison among these existing methods for the  
70 general case of discrete networks can be found in Veenendaal and Block (2012).

71 For both free-form and form-found grid shells, the state-of-the-practice in design is to choose  
72 the global shape of the structure first. Nodal connectors are typically considered last, with their ge-  
73 ometry dictated by the form and sized to meet demands. This results in connections that are unique,  
74 inefficient, and difficult to fabricate. Each structure is designed as one-of-a-kind, leading to further  
75 inefficiencies. In contrast, this research advocates for a new paradigm in design: start with connec-

76 tors that are designed for ease of fabrication and erection, and then develop an efficient structural  
77 form that is consistent with these connectors. This approach “modularizes” the nodal connector  
78 such that identical connectors can be used repeatedly throughout the structure and among many  
79 structures. This research specifically introduces a “modular connector” for grid shell structures  
80 which is a prefabricated, steel nodal connector that is designed for ease of fabrication (Figure 1A).  
81 Standard rolled wide flange members serve as members, joined by modular connectors, to achieve  
82 free-form or form-found steel grid shells (Figure 1C and D). Specifically, the modular connector is  
83 comprised of top and bottom flange plates, as well as web plates, all cut from flat steel plate. The  
84 components are welded to one another and to a standard round hollow structural section (HSS) at  
85 the center. The open geometry of the connector facilitates the welding process. The modular con-  
86 nector includes straight starter segments for connection to the wide flange members. The flanges  
87 and webs of the connector and the members are joined independently through bolted splice con-  
88 nections in double shear (Figure 1B). This results in a moment-resisting connection that enables  
89 truss-like, membrane-like, or beam-like behavior and provides enhanced resiliency as member loss  
90 can be tolerated. Using only bolted connections (i.e., no field welding) provides savings in erec-  
91 tion costs. Various structural forms (to improve efficiency or to satisfy architectural constraints)  
92 are achieved by using bent flange splice plates, which can be the adjustable bolted steel plate con-  
93 nection (Gerbo et al., 2016, 2018, 2020a,b). These bent flange plates enable the components to be  
94 joined at an angle,  $\gamma$ . The modular connector and wide flange members are also designed for ease  
95 of transportation in standardized by the International Standard Organization shipping containers  
96 (ISO containers, hereafter). Ultimately, this paper proposes a new kit-of-parts for the design and  
97 fabrication of a wide variety of steel grid shells comprised of (1) modular connectors, (2) standard  
98 wide flange sections, and (3) bolted splice connections.

99 Tumbava et al. (2021) proposed an analogous, but different, approach for the modular construc-  
100 tion of steel bridges. In Tumbava et al. (2021), a two-dimensional (2D) modular joint is proposed  
101 which is comprised of cold bent flange plates, flat flange plates, and a web (Figure 2A). This mod-  
102 ular joint replaces the function of a gusset plate with a more reliable connection and enables the

103 upper chord, lower chord, and diagonal members to be standard rolled wide flange sections (Fig-  
104 ure 2B). The modular connector introduced in this paper, as well as the modular joint proposed in  
105 Tumbeva et al. (2021), represent a new paradigm in modular construction, in which the connection  
106 between members becomes the module and structural forms are derived to be consistent with the  
107 modules.

108 Extensive research has been performed on developing and investigating the behavior of nodal  
109 connectors for grid shells and spatial structures, with a particular focus on single layer grid shells.  
110 For example, Feng et al. (2015) experimentally and numerically investigated the behavior of mod-  
111 ified conventional bolted joints for cable-braced grid shells with a particular focus on the joint  
112 stiffness and failure mechanism. Seifi et al. (2018) proposed two different approaches for simpli-  
113 fying the design and analysis of nodal connectors for grid shells that have improved mechanical  
114 properties and can be 3D printed. These approaches are more focused on the design of nodal  
115 connectors that are specific to a project, as opposed to promoting a modular kit-of-parts. Ma  
116 et al. (2016a,b) proposed a new semi-rigid bolt-column joint (BC), comprised of a hollow column  
117 node, bolts, and a cone part for connection to rectangular hollow or wide flange sections. The BC  
118 joint achieves the geometry of the grid shell by positioning the entire joint at a desired angle. Oh  
119 et al. (2016) numerically investigated a new hollow spherical connector - FREE node - to deter-  
120 mine an efficient node shape while considering the connector's failure mode and stiffness. The  
121 FREE node is comprised of hollow sphere, sub-wings, and wings that can connect hollow sections  
122 through bolting. To achieve different angles however, welding of the members directly to the hol-  
123 low sphere may be required. Many patents have been developed for various universal connections  
124 [e.g. Rochas (2014), Allred et al. (2013), Boots (2008), Reynolds et al. (2006)]. In comparison  
125 to existing technologies, the connector proposed in this paper (1) enables members to be joined  
126 at different angles thus achieving wide variability in structural form, (2) can join many different  
127 standard wide flange sections for high efficiency and adaptability for demand, and (3) facilitates a  
128 more reliable, moment-resisting connection for enhanced resiliency.

## 129 OBJECTIVES AND SCOPE

130 This research addresses a major challenge in the design, fabrication, and erection of grid shells:  
131 nodal connectors. This research introduces a novel modular connector that retains the advantages  
132 of modularity (i.e., that a single, prefabricated component can be repeated throughout a structure  
133 and among many structures, leading to cost savings and time efficiencies) while being able to  
134 achieve free-form and form-found grid shells using standard sections as members and providing  
135 enhanced resiliency through moment-resisting connections. The specific objectives of this paper  
136 are to: (1) develop the geometry of the modular connector for free-form and form-found grid shells,  
137 (2) propose a form-finding methodology to achieve rational, compression-only grid shells that meet  
138 geometric and structural criteria, and (3) demonstrate the promise of this approach through finite  
139 element (FE) numerical analyses. Ultimately this research presents a new paradigm in design in  
140 which nodal connectors are developed for ease of design, fabrication, and erection and structural  
141 forms, consistent with the constraints of the nodal connector, are chosen.

## 142 GEOMETRIC PARAMETERS OF THE MODULAR CONNECTOR

143 To develop the modular connector, the following parameters are defined (Figure 3): (1) number  
144 of starter segments,  $ns$ , (2) angle between starter segments,  $\theta$ , (3) depth of the web plate,  $w$ , (4)  
145 flange width,  $f$ , (5) radius,  $R$ , and (6) starter segment length,  $d$ .

146 This research proposes a modular connector with six starter segments ( $ns = 6$ ) to be able  
147 to join up to 6 members, thus generating a triangulated grid shell which is desirable to ensure  
148 global stability of the structure. Triangulation also enables flat planes of glass to be used between  
149 members to form the enclosed space. Analogous modular connectors with different number of  
150 starter segments could also be considered. With six starter segments and the choice that the starter  
151 segments are evenly spaced, the angle between starter segments,  $\theta$  is then  $60^\circ$ .

152 The depth of the web plate,  $w$  is selected such that the modular connector can join a range of  
153 standard wide flange sections, thereby allowing greater variability in structural forms. For standard  
154 wide flange members with the same nominal depth (i.e., sections with the same WXX designation,  
155 where XX refers to the nominal depth), the web depth (i.e., the section depth minus the thickness

156 of the two flanges) is approximately the same as a result of the steel rolling fabrication process.  
157 This paper specifically focuses on a set of ten W14 wide flange sections between W14x109 and  
158 W14x257 (AISC, 2011). Therefore, the depth of the web plate,  $w$  of the modular connector is  
159 selected to be 320 mm (12.6 in.) that is the average web depth of the selected set of wide flange  
160 sections. A designer could choose other depths corresponding to other types of wide flange sec-  
161 tions. Alternatively, fill plates could be used. Note that, when bent flange splice plates are used, fill  
162 plates are required as the modular connectors and members are connected at an angle,  $\gamma$  (Figure  
163 1B). The modular connector flange width,  $f$  is 406 mm (16 in.), which is selected to be on the  
164 upper limit of the flange widths of the considered wide flange sections.

165 The radius,  $R$  is chosen to be 508 mm (20 in.). A large radius is beneficial as it reduces stress  
166 concentrations at the curved region of the flange plates of the connector as well as increases the  
167 cross sectional area of the flange. The starter segment length,  $d$  is chosen to be 229 mm (9 in) as  
168 a minimum value to facilitate a bolted splice connection. The length of each of the web plates,  $e$   
169 can be calculated as follows:

$$e = d + \frac{(f/2) + R}{\tan \frac{\theta}{2}} - c \quad (1)$$

170 where  $c$  is the radius of the HSS section including its wall thickness.

171 The thickness of the flange plates and the web plates as well as the size of the HSS section can  
172 be determined based on demand.

## 173 **FREE-FORM GRID SHELLS**

174 Free-form grid shells can be approximately achieved (i.e., any polygonalized version of the  
175 desired continuous shape) using the modular connector, wide flange members, and bolted splice  
176 plate connections. To achieve variability in depth, this approach relies on angle changes,  $\gamma$  between  
177 modular connectors and wide flange sections about a single axis orthogonal to the webs which is  
178 achieved through bent flange splice plates (Figure 1B). Angles about any other axis would be  
179 incompatible with this system. Thus, it is required that all modular connectors be parallel to one

180 another. It can be envisioned that all modular connectors are projected onto a parallel, planar ( $x-y$ )  
181 mesh to ensure this requirement (Figure 4A). For the  $\theta = 60^\circ$  modular connector with 6 starter  
182 segments, this mesh would be comprised of equilateral triangles (Figure 4A). Based on the desired  
183 structural form, the  $z$ -coordinate of each modular connector, relative to the planar ( $x-y$ ) mesh,  
184 is determined (Figure 4B). The designer can select the mesh size,  $h$  of the equilateral triangles in  
185 the planar mesh, and thus, the  $x-y$  coordinates of the modular connectors, based on aesthetics,  
186 architectural constraints, and/or structural demand. Each of the modular connectors join wide  
187 flange members through splice plates (bent or flat). If the modular connectors are placed at every  
188 vertex, the grid shell is triangulated, which provides stability (without relying on the moment-  
189 resisting connections) and enables flat glass to be used between members to enclose the space. The  
190 moment-resisting connection could then be relied upon in the event of member loss or replacement.  
191 A designer need not place modular connectors at every vertex of the mesh or use members between  
192 every modular connector, thereby providing freedom for openings or architectural vision (Tumbeva  
193 et al., 2018).

## 194 **PROPOSED METHODOLOGY FOR FORM-FOUND GRID SHELLS**

195 While any free-form undulating structure can be achieved, a structure that carries primarily  
196 axial load is preferred from a structural efficiency perspective, as opposed to bending dominant be-  
197 havior. This paper proposes a new methodology for achieving form-found grid shells that primarily  
198 carry axial compression and are consistent with the modular connector.

199 In the state-of-the-practice, designers typically select a form-finding technique depending on  
200 the structural material. The boundaries of the structure, boundary conditions, and external loads are  
201 prescribed, as well as an initial geometry (or mesh). The form is then found with little regard to the  
202 geometry of the nodal connections between elements. Often unique nodal connection geometries  
203 result, leading to the previously discussed challenges in design, fabrication, and erection.

204 Instead, this research advocates for using a form-finding technique that is well-suited to con-  
205 nections that are easy to fabricate and erect. In this case, the focus is on the modular connector.  
206 TNA is well-suited for grid shells comprised of modular connectors as it derives a compression-



207 only (under one load scenario) grid shell,  $G$  from an equilibrated planar mesh,  $\Gamma$  and the positions  
208 of the “free” nodes are found only in a direction perpendicular to that mesh (Figure 5). The  $x - y$   
209 coordinates of these free nodes are defined by the initial mesh and thus, are not found. The “sup-  
210 port” nodes represent boundary conditions, with their  $x - y - z$  coordinates being prescribed by  
211 the user. Truss elements are used to span between nodes. Truss elements are used to span between  
212 nodes. The geometry of the modular connectors is neglected (i.e., they are simplified to be in-  
213 finitesimal nodes that connect truss elements). In the general formulation of TNA, the planar mesh  
214 is generated by the user (Block and Ochsendorf, 2007; Block, 2009). As shown in Figure 5, the  
215 planar mesh in this research is the mesh of equilateral triangles that is necessary for the modular  
216 connector, as discussed in the prior section. For a user-defined mesh size,  $h$  and loading, the  $z$ -  
217 coordinate of the free nodes is found. Thus, the planar coordinates of the modular connector are  
218 retained and the structural depth is found. This approach is limited to extrusions from a flat plane,  
219 where all modular connectors are parallel to this flat plane (i.e., the equilibrated planar mesh,  $\Gamma$ ).

220 This paper proposes a methodology for finding efficient (i.e., minimized weight) compression-  
221 only grid shells that are compatible with the modular connector and meet structural constraints on  
222 behavior as well as geometric requirements (Figure 6). Minimizing the self-weight also results  
223 in a system that is sufficiently light for ease of transportation and erection. Alternatively, other  
224 objectives could be considered such as fabrication or erection cost. The methodology could also  
225 be reformulated as a multi-objective optimization problem that includes material efficiency and  
226 cost. For a user-defined span length,  $S$ , the methodology investigates  $n$  number of different planar  
227 meshes comprised of equilateral triangles with mesh size,  $h$ . An iterative TNA procedure is used  
228 to find the form of the compression-only grid shell and evaluate the axial forces in the member. To  
229 develop the geometry of the structure, TNA relies on a scale factor,  $\xi$ , to be discussed later. For  
230 each combination of  $h$  and  $\xi$ , the iterative TNA procedure is carried out. The feasibility of each  
231 form is then evaluated through structural and geometric constraints. One lowest weight solution,  
232 corresponding to one value of  $\xi$ , will result for each mesh size,  $h$ . The methodology ultimately  
233 finds the lowest weight solution among all of the different mesh sizes,  $h$ . While Figure 6 gives an

234 overview of the proposed methodology, Figure 7 provides a detailed flow-chart of this methodology  
235 for one value of  $h$  and  $\xi$ .

236 Within the proposed methodology, an iterative TNA procedure is performed (dark shaded re-  
237 gion in Figure 7) to find a compression-only grid shell form (i.e., the  $z$ -coordinates of the free  
238 nodes) and calculate the axial forces in the members. TNA, as described in the next section, is able  
239 to find a compression-only form under a single, known load. In this research, the load includes the  
240 self-weight of the wide flange members, the self-weight of structural glass panels between mem-  
241 bers, and a snow load. The geometry and self-weight of the modular connectors is ignored for  
242 simplicity. A major contribution to the overall load and thus, the form of the final 3D grid shell,  
243 is the self-weight. However, this weight cannot be known until a form is found, as it is calculated  
244 based on the structure's geometry which changes depending on the load. Thus, this research uses  
245 an iterative approach in which the load and form are updated until convergence is achieved. Con-  
246 vergence is defined as the difference in the  $z$ -coordinates of each free node between two successive  
247 iterations being less than a user-defined limit. Note that, the starter segments of the modular con-  
248 nector introduce an eccentricity of loading. The methodology neglects this eccentricity, as well  
249 as the moment resisting connections between members, for the purpose of rapidly converging on  
250 form. However, it would need to be incorporated for design.

251 Prior to starting the iterative approach, a database of  $M$  number of standard wide flange sections  
252 is defined and the section size,  $B_a$ , where  $a = 1$  ( $a \leq M$ ), with the smallest weight per linear foot  
253 is selected (Figure 7). Then, the iterative approach begins by setting the  $z$ -coordinates of the  
254 free nodes,  $z_f^i$  to zero, where  $i$  is the iteration number and  $f$  refers to the free nodes. Using the  
255 coordinates  $z_f^i$ , the load,  $P^i$  is calculated. TNA is then used to find a new set of nodal coordinates  
256  $z_f^{i+1}$ . To avoid an infinite number of iterations and to allow a form to be found, an upper limit on  
257 the nodal coordinates is imposed:  $z_m = S/5$ . If  $z_f^{i+1}$  exceeds the limit  $z_m$ , a form is not found  
258 for that combination of  $h$  and  $\xi$ . If  $z_f^{i+1} \leq z_m$ , the convergence criteria:  $z_f^{i+1} - z_f^i \leq 0.001$ , is  
259 then evaluated for each node. If convergence is not achieved, the magnitude of the load is updated  
260 based on the found form and another iteration is performed. If convergence is satisfied, the form of

261 the grid shell is found and the methodology continues with evaluating the structural and geometric  
 262 constraints (light gray shaded region in Figure 7)

263 To ensure stability, the methodology includes structural constraints related to member buckling  
 264 and global system buckling. The member buckling constraint requires that the axial force,  $F$  in  
 265 each member must be less than the compressive strength,  $U$  of that member (for its weak axis)  
 266 expressed in vector form:  $\vec{F} < \vec{U}$ , where  $U$  is calculated by (AISC, 2011):

$$U = \phi S_{cr} A_g \quad (2)$$

267 where  $\phi$  is the resistance factor ( $\phi = 0.9$ ),  $A_g$  is the cross sectional area of the member, and  $S_{cr}$  is  
 268 the critical stress determined as:

$$S_{cr} = \begin{cases} \left[ 0.658 \frac{F_y}{F_e} \right] F_y & \text{When } \frac{KL_u}{r_g} \leq 4.71 \sqrt{\frac{E}{F_y}} \\ 0.877 F_e & \text{When } \frac{KL_u}{r_g} > 4.71 \sqrt{\frac{E}{F_y}} \end{cases} \quad (3)$$

269 where  $F_y$  is the yield strength [ $F_y = 345$  MPa (50 ksi)],  $E$  is the modulus of elasticity [ $E = 200$  GPa  
 270 (29000 ksi)],  $K$  is the effective length factor ( $K = 1$ ),  $r_g$  is the radius of gyration about the weak  
 271 axis of the wide flange member, and  $L_u$  is the unbraced length of the member. While the geometry  
 272 of the modular connector is not modeled in the TNA analysis (truss elements are assumed to span  
 273 between nodal coordinates), the unbraced length is taken as the actual length of the wide flange  
 274 member between what would be modular connectors,  $L_m$  (Figure 1C).  $F_e$  is the elastic buckling  
 275 stress calculated by:

$$F_e = \frac{\pi^2 E}{\left( \frac{KL_u}{r_g} \right)^2} \quad (4)$$

276 If the member buckling constraint is satisfied, the global system buckling constraint is evalu-  
 277 ated, which requires that the smallest critical buckling load factor,  $\lambda_1$ , exceeds 1.  $\lambda_1$  is determined

278 by solving the generalized eigenvalue problem  $[\bar{K}_E + \lambda\bar{K}_G] \vec{\psi} = 0$ , in which  $\bar{K}_E$  is the global  
279 linear elastic stiffness matrix,  $\bar{K}_G$  is the global geometric stiffness matrix, and  $\vec{\psi}$  is the eigenvec-  
280 tors. In the buckling analysis, the 3D structure is modeled with truss elements and the member  
281 axial forces calculated through the TNA iterative approach are used. The boundary conditions at  
282 the supported nodes are: translation restrained in  $x$ ,  $y$ , and  $z$  directions, free rotation. If either  
283 structural constraint is not met, the section size is increased and the iterative TNA procedure is re-  
284 peated. If the maximum section size (i.e.,  $a=M$ ) is investigated and still fails to meet the structural  
285 criteria, the methodology ends with no feasible solution for that combination of  $h$  and  $\xi$ .

286 Note that, although the methodology uses standard wide flange sections, it is also compatible  
287 with other section types (e.g., L-sections, hollow structural sections). However, if the sections are  
288 slender, Equation 3 would need to be modified according to AISC (2011). The global system buckling  
289 constraint would not be affected by the change in section type as it assumes truss elements and the  
290 axial forces are determined prior to solving the eigenvalue problem.

291 If the structural constraints are met, a geometric constraint, related to transporting the kit-of-  
292 parts in an ISO shipping container with inner length,  $T = 12$  m (39 ft 4 in.) is evaluated (ISO,  
293 2013). The length of each member,  $L_m$  is required to be less than  $T$ . If this constraint is satisfied,  
294 a second geometric constraint, related to the adjustable bolted steel plate connection developed by  
295 Gerbo et al. (2016, 2018, 2020a,b), is evaluated. The behavior of the connection, comprised of  
296 flat or prebent flange plates that are bent further in the field via bolt tightening, was investigated  
297 for angle changes up to  $35^\circ$ . Thus, the angles between the modular connectors and wide flange  
298 members,  $\gamma$  are limited to  $\gamma_m = \pm 35^\circ$ . If either of the two geometric constraints is not met, then  
299 the methodology ends with no feasible solution for that combination of  $h$  and  $\xi$ .

### 300 **Form-Finding and Analysis through TNA**

301 This research uses TNA to find compression-only forms and to calculate the axial forces in  
302 the members. TNA, proposed by Block and Ochsendorf (2007) and Block (2009), is a 3D version  
303 of thrust line analysis developed for form-finding and analysis of masonry and concrete vaulted  
304 structures. The form (i.e., the  $z$ -coordinates of the free nodes) of a compression-only vault is

305 found through planar ( $x - y$ ) equilibrium (via reciprocal form,  $\Gamma$  and force,  $\Gamma^*$  diagrams, to be  
 306 discussed later) and perpendicular ( $z$ -direction) equilibrium (Figure 5). The vault is modeled as a  
 307 discrete network of truss elements with loads applied only at the nodes, in the  $z$ -direction. Each  
 308 truss element is required to be in compression only. Based on TNA, Rippmann (2016) proposed a  
 309 framework for the interactive design of funicular shell structures which resulted in the development  
 310 of the digital design tool RhinoVAULT (Rippmann et al., 2012). This section presents a brief  
 311 review of TNA and how it is used in this research, which unless stated otherwise, is adapted from  
 312 Block (2009).

313 In this research, TNA is used to determine the  $z$ -coordinates of the free nodes of a three-  
 314 dimensional (3D) compression-only grid shell,  $G$  for one loading condition, applied as point loads  
 315 in the  $z$ -direction (Figure 5). The  $z$ -coordinates are found through evaluating the perpendicular  
 316 ( $z$ -direction) equilibrium of the structure,  $G$  defined as follows:

$$\bar{C}_f^T (\bar{L}_{xy}^{-1} \bar{F}_{xy}) \bar{C} \vec{z} - \vec{p}_z = 0 \quad (5)$$

317 where  $\bar{C}$  is an  $m \times n$  matrix that represents the connectivity between  $m$  number of members and  $n$   
 318 number of nodes as follows:

$$\bar{C}_{mn} = \begin{cases} +1 & \text{if member } m \text{ starts with higher index node } n \\ -1 & \text{if member } m \text{ ends with lower index node } n \\ 0 & \text{all other entries} \end{cases} \quad (6)$$

319 The matrix  $\bar{C}$  can be divided into two parts:  $\bar{C}_f$  and  $\bar{C}_s$  corresponding to the free ( $f$ ) and support  
 320 ( $s$ ) nodes, respectively. In Equation 5,  $\bar{L}_{xy}$  is a diagonalized matrix containing the lengths of the  
 321 projection of the truss elements (with lengths  $\bar{L}$  in 3D) onto the  $x - y$  plane (Figure 5).  $\bar{F}_{xy}$  is a  
 322 diagonalized matrix containing the  $x - y$  component of the axial force in the members which is

323 found through the planar ( $x - y$ ) equilibrium of the structure. It is calculated as follows:

$$\bar{F}_{xy} = \bar{L}_{xy}^* \xi \quad (7)$$

324 where the matrix  $\bar{L}_{xy}^*$  is discussed in the following subsection. In Equation 5,  $\vec{p}_z$  is a vector of the  
325 applied loads.

326 Equation 5 can then be solved for the  $z$ -coordinates of the free nodes,  $\vec{z}_f$ :

$$\vec{z}_f = \bar{D}_f^{-1} \left( \frac{1}{\xi} \vec{p}_z - \bar{D}_s \vec{z}_s \right) \quad (8)$$

where  $\vec{z}_s$  is the vector of the  $z$ -coordinates of the support nodes and the matrices  $\bar{D}_f$  and  $\bar{D}_s$  are:

$$\bar{D}_f = \bar{C}_f^T (\bar{L}_{xy}^{-1} \bar{L}_{xy}^*) \bar{C}_f \quad (9)$$

$$\bar{D}_s = \bar{C}_s^T (\bar{L}_{xy}^{-1} \bar{L}_{xy}^*) \bar{C}_s \quad (10)$$

327 The diagonalized matrix,  $\bar{F}$  containing the axial forces in each member is calculated by:

$$\bar{F} = \bar{L} \bar{L}_{xy}^{-1} \bar{L}_{xy}^* \xi \quad (11)$$

### 328 **Reciprocal Form and Force Diagrams**

329 The planar ( $x - y$ ) equilibrium in TNA is solved through reciprocal form,  $\Gamma$  and force,  $\Gamma^*$   
330 diagrams (Figure 5).

331 Reciprocal diagrams have mainly been used to design and analyze statically determinate 2D  
332 funicular structures under loads applied at the nodes. For 2D structures, the form diagram,  $\Gamma$   
333 is a funicular polygon representing the structure's geometry, with nodes that are numbered and  
334 spaces between members represented by capital letters (Figure 8). Its reciprocal force diagram,  
335  $\Gamma^*$  graphically represents the internal forces in the members, where the length of each branch  
336 corresponds to a scaled magnitude of the force in the parallel member of the form diagram. For

337 example, the magnitude of the force in the member connecting nodes 2 and 8 (between spaces A  
338 and B) in Figure 8A is related to the length of the a-b branch of the force diagram by the scale  
339 factor. Each of the closed polygons in the force diagram represents the static equilibrium of a  
340 node in the form diagram (Maxwell, 1864). The concept of reciprocal diagrams is extended for 3D  
341 structures in TNA by Block and Ochsendorf (2007) and Block (2009).

342 In TNA, the form diagram,  $\Gamma$  is the planar mesh from which the 3D structure,  $G$  is found  
343 (Figure 5). The planar equilibrium of  $\Gamma$  and thus, the planar equilibrium of  $G$ , is represented by  
344 the force diagram,  $\Gamma^*$ . The  $x - y$  component of the axial force in the members,  $\bar{F}_{xy}$  is calculated  
345 from Equation 7, where  $\bar{L}_{xy}^*$  (Figure 5) is the length of the branches of the force diagram and  $\xi$  is  
346 the scale factor defined as force per length of the force diagram. This is used in TNA for solving  
347 the perpendicular equilibrium.

348 In this research, the form diagram,  $\Gamma$  is the planar mesh of equilateral triangles that is required  
349 for the modular connector. Depending on the span length,  $S$  and mesh size,  $h$ , there are two types  
350 of form diagrams:  $\Gamma_1$  and  $\Gamma_2$  (Figure 9). Both diagrams are developed by calculating the  $q$  number  
351 of inner members (solid grey lines in Figure 9) along the span length by:  $q = S/h$ , where  $q$  is  
352 rounded down to the next even integer. The length of the two boundary members along the span  
353 length, connecting nodes 5 and 6 respectively in  $\Gamma$  is:  $b = (S - qh)/2$ . To maintain the planar  
354 mesh of equilateral triangles, the number of inner members along lines 1-3, 2-4, 5-6 are the same  
355 and equal to  $q$ . If  $q/2$  is even, then the form diagram is  $\Gamma_1$ . If  $q/2$  is odd, the form diagram is  $\Gamma_2$ .

356 For either type of form diagram, an infinite number of force diagrams exist, each representing  
357 a state of planar equilibrium of the 3D structure and therefore, a unique grid shell form. Figure 9  
358 shows the force diagrams,  $\Gamma_1^*$  and  $\Gamma_2^*$  developed for form diagrams,  $\Gamma_1$  and  $\Gamma_2$ , respectively. There  
359 are different ways to obtain the force diagram such as drawing it manually, using optimization  
360 methods (Block, 2009), or through iteration and additional constraints as proposed by Rippmann  
361 (2016). In this research, both force diagrams are constructed in a similar way by starting with  
362 the center node of the form diagram. The force diagram is developed in a clockwise direction to  
363 ensure that all members are in compression. As this research aims for a modular approach where

364 all members are the same section size, it is desirable that all members have the same magnitude  
365 of force. To achieve this, the inner branches in the force diagram (shown with solid grey line in  
366 Figure 9) are required to have an equal length.

367 This research also employs a tension ring at the boundary of the grid shell (shown as black  
368 dashed lines in Figure 9) to counteract the horizontal thrust that would be generated at the sup-  
369 ports. The tension ring is represented in the force diagram,  $\Gamma^*$  by branches that connect the center  
370 node with each of the edge nodes. The parallel tension ring members in the form diagram,  $\Gamma$  are in-  
371 tersecting the boundary members at the support nodes. Thus, the length of the boundary members,  
372 except those connecting nodes 5 and 6, in the form diagram,  $\Gamma$  are determined from the polygons  
373 constructed by the boundary and tension ring members. For the force diagram,  $\Gamma^*$  to be comprised  
374 of closed polygons, the length of the boundary branches must be twice that of the inner branches.  
375 Note that in the force diagram type  $\Gamma_1^*$ , the length of the boundary branches corresponding to the  
376 boundary members with dashed light grey lines in the form diagram,  $\Gamma_1$  is the same as the length  
377 of the inner branches (Figure 9A). In the force diagram,  $\Gamma^*$ , the length of the corner branches cor-  
378 responding to the corner members in the form diagram,  $\Gamma$  (connecting nodes 1, 2, 3, and 4, shown  
379 with solid grey lines in Figure 9) is the same as the length of the inner branches. To form closed  
380 polygons, the edge branches in the force diagram,  $\Gamma^*$  intersect the boundary branches, thus, the  
381 length of the edge branches is found from the geometry of the constructed polygons.

382 As mentioned earlier, the force diagram,  $\Gamma^*$  is a graphical representation of the forces, with the  
383 length of a branch indicating the magnitude of the force. The factor  $\xi$  is the scale that relates the  
384 length of a branch to a force magnitude (Equation 7). Because the coordinates  $z_f$  are inversely  
385 proportional to  $\xi$ , a smaller scale factor  $\xi$  results in higher values of  $z_f$  and thus, a deeper grid shell  
386 form,  $G$  (Equation 8). For the same form diagram and load, by only varying the scale factor,  $\xi$ ,  
387 an infinite number of grid shell forms can be developed. The inner branches of the force diagram,  
388  $\Gamma^*$  are given a unit length and all other branches are related to that unit length as discussed above.  
389 The proposed form finding methodology considers a range of scale factors,  $\xi$  to be able to select  
390 an efficient (i.e., minimized material use) grid shell from a vast number of solutions, based on



391 structural performance criteria as well as transportation requirements.

## 392 CASE STUDIES

393 The developed methodology is demonstrated through four case studies with span lengths:  $S_1$   
394 = 42.7 m (150 ft),  $S_2 = 53.3$  m (175 ft),  $S_3 = 61$  m (200 ft), and  $S_4 = 76.2$  m (250 ft). To explore  
395 a wide range of forms, the mesh size,  $h$  varies between 6.1 m (20 ft) and  $h_{max}$  in increments of  
396 0.305 m (1 ft), where  $h_{max} = T + 2(e + c) = 15.1$  m (49.7 ft). The limit,  $h_{max}$  is based on the  
397 requirement that each member be transported in an ISO container (with inner length,  $T$ ), including  
398 also the dimensions of the joint. The scale factor,  $\xi$  ranges from 1 to 1,000 in increments of 10.  
399 Hence, a total of 3,000 possible solutions were investigated for each span. The database of standard  
400 wide flange sections includes all section sizes from W14x109 through W14x257 resulting in  $M =$   
401 10 (AISC, 2011). It is assumed that all members have the same section to promote modularity.

402 Two load combinations as per the American Society of Civil Engineers Minimum Design  
403 Loads and Associated Criteria for Building and Other Structures Standard ASCE 7-16 (ASCE/SEI,  
404 2017) were considered:  $I_1 = 1.4D$  and  $I_2 = 1.2D + 1.6N$ , where  $D$  is the dead load and  $N$  is  
405 the snow load. For each iteration of the TNA procedure, the larger of the two load cases was  
406 applied. The dead load,  $D$  includes the self-weight of the wide flange members and self-weight  
407 of a structural glass of  $0.781$  kN/m<sup>2</sup> (15 psf) which is assumed to span between members (AISC,  
408 2011). The self-weight of the modular connector is not included for simplicity. The magnitude  
409 of the snow load,  $N$  is  $0.666$  kN/m<sup>2</sup> (13.9 psf), calculated per the design code assuming that the  
410 structure is a flat roof (ASCE/SEI, 2017). The snow load is calculated based on the projected area  
411 of the structure. Both the dead and snow loads are applied as point loads at the free nodes.

412 For each span length, the grid shell with the lowest weight is shown in Figure 10. Spans  $S_2$ ,  $S_3$ ,  
413 and  $S_4$  have similar forms, while span  $S_1$  requires less number of members to achieve the desired  
414 span length. Table 1 contains summarized results including: the lowest weight,  $W_{min}$ , mesh size,  
415  $h$ , the  $z$ -coordinate of the center node,  $z_c$ , largest axial force,  $F_{max}$ , member buckling capacity  
416 factor,  $\mu = \max(F/U)$ , member section size, and the lowest eigenvalue,  $\lambda_1$ .

417 For all four spans, the methodology selects two of the largest section sizes in the database,

418 W14x233 for  $S_1$  and W14x257 for the other three. This is primarily to satisfy the global system  
419 buckling constraint of  $\lambda_1 \geq 1$ . Both section sizes have a large stiffness in their weak axis resulting  
420 in a significant compressive capacity in each member, while the axial forces in the members are  
421 much lower in comparison. This is indicated by the lower value of the member buckling capacity  
422 factor,  $\mu$ .

423 To further demonstrate how the proposed methodology finds the grid shell with the lowest  
424 weight, the change in  $z_c$  based on self-weight,  $W$ , peak axial force,  $F_{max}$ , critical buckling load  
425 factor,  $\lambda_1$ , and member buckling capacity factor,  $\mu$  is traced for different values of the scale factor,  
426  $\xi$  for the case study  $S_2 = 53.3$  m (175 ft) and mesh size  $h = 9.14$  m (30 ft) in Figure 11. It is evident  
427 that deeper grid shells have a higher structural weight as the length of the truss elements increase  
428 (Figure 11A). However, higher values of  $z_c$  lower the peak axial force and thus, increase the global  
429 system buckling capacity as  $\lambda_1$  becomes considerably higher than the requirement (Figure 11B  
430 and C). The results in Figure 11D indicate that the member buckling constraint does not govern  
431 the selection of the grid shell form, as the factors,  $\mu$  for each value of  $\xi$  are significantly lower than  
432 1. For shallower grid shells, the largest section size is selected and a low value of  $\mu$  results. As  
433  $z_c$  increases, axial forces decrease and smaller section sizes are selected with  $\mu$  also increasing,  
434 until the smallest section size of W14x109 is chosen. From there, with increasing  $z_c$ , the member  
435 forces and thus, also  $\mu$  continue to decrease. Overall, to minimize the weight of the structure, the  
436 methodology selects a form that has a low structural depth and the largest wide flange section size  
437 to achieve  $\lambda_1$  close to the requirement of 1.

438 For span  $S_2$ , a total of 100 scale factors  $\xi$  were investigated. However, only 27, in the range  
439 between 80 and 300, were found to result in feasible solutions, as shown in Figure 11. The lower  
440 bound of the factor  $\xi$  is primarily controlled by geometric constraints as deeper grid shells increase  
441 both the angles,  $\gamma$  and the member lengths,  $L_m$ . The upper bound of  $\xi$ , however, is governed by  
442 the structural constraints, as too shallow of grid shells significantly lower both member and global  
443 system buckling capacity.

444 From all 31 mesh sizes,  $h$  investigated for each span length, not all result in feasible solutions.

445 Figure 12 shows the mesh sizes,  $h$  that result in feasible solutions for all span lengths. The reduced  
446 number of mesh sizes is primarily due to the system buckling constraint as grid shells with a  
447 form diagram type  $\Gamma_2$  resulted in a significantly low eigenvalues  $\lambda_1$ , and thus insufficient buckling  
448 capacity.

#### 449 **DEMONSTRATION OF FEASIBILITY**

450 Grid shell structures can experience different buckling modes including member buckling, node  
451 buckling (snap-through buckling), torsional buckling of the node, and global buckling (Gioncu,  
452 1995). The nodal connector, particularly in single layer grid shells, has a significant importance  
453 on the global stability of the structure. There is extensive research on buckling of grid shells  
454 and other space frame structures that takes into account the nodal connector, initial imperfections,  
455 and geometric and material nonlinearity. Suzuki et al. (1992) investigated the effect of material and  
456 geometric nonlinearity on the buckling behavior of single layer reticulated domes with rigidly con-  
457 nected members. Kato et al. (1998) studied the collapse behavior of domes with semi-rigid nodal  
458 connectors, incorporating initial imperfections. The collapse mechanism of single layer domes is  
459 significantly influenced by both the geometry of the structure and the connector's stiffness. Lopez  
460 et al. (2007a) proposed an approach to determine the buckling load of domes with semi-rigidly  
461 connected members for various loading condition and geometric parameters. Additionally, Lopez  
462 et al. (2007b) conducted numerical and experimental studies in which the node stiffness, member  
463 properties, and load distribution were incorporated to determine the critical buckling load. Hwang  
464 (2010) numerically investigated the impact of the connector's stiffness and initial imperfections on  
465 the buckling of the grid shells. The study showed that the structure can experience failure under  
466 relatively low loads as a result of bending stresses which lower the stiffness of the joint.

467 Given the importance of global stability in grid shell design, this research demonstrates the  
468 feasibility of the modular connector and the developed forms through 3D FE linear eigenvalue  
469 buckling analyses in the software package ABAQUS/Standard (ABAQUS, 2016) (Figure 13). The  
470 modular connectors and the wide flange sections are modeled using 4-node (S4R) or 3-node (S3R)  
471 reduced integration general purpose shell elements with six degrees of freedom. A 50.8-mm (2-

472 in.) mesh size, determined through mesh refinement studies, was used for all components. The  
473 tension tie is not modeled for simplicity. Instead, the outer wide flange members are connected to  
474 horizontal wide flange segments, representing what could be a final boundary modular connector.  
475 At the end of each of the wide flange segments at the node intersecting the web and bottom flange,  
476 the following boundary conditions are applied: translation is restrained in all three orthogonal  
477 directions, free rotation in all directions. The bolted splices are not modeled explicitly. Instead  
478 all nodes along the edges of the flanges of the members are tied to the nodes along the edges  
479 of the flanges of the modular connectors or the horizontal wide flange segments. Similarly, the  
480 nodes along the web edge of the members are tied to the nodes along the web edge of the modular  
481 connector or the horizontal wide flange segments. The constraint ties all degrees of freedom at the  
482 nodes throughout the duration of the analysis. Grade 50 structural steel properties are assumed:  
483 345 MPa (50 ksi) minimum yield strength, 200 GPa (29000 ksi) modulus of elasticity, 7850 kg/m<sup>3</sup>  
484 (490 lbs/ft<sup>3</sup>) steel density, and 0.3 poisson's ratio.

485 The linear eigenvalue buckling analysis was performed for case study  $S_2 = 53.3$  m (175 ft) and  
486 load scenario  $I_2$  (highest magnitude of the load). In this model, the self-weight of all structural  
487 steel components was applied through the specific density and the acceleration of gravity. Both,  
488 self-weight of the structural glass and snow load were applied as a point load at every elemental  
489 node along the middle line of the top flange of the wide flange members. Members were modeled  
490 using the W14x257 section size determined earlier through the proposed methodology. Based on  
491 this section size, the thickness of the flange plates as well as the thickness of the web plates in the  
492 modular connectors were selected to be the same as for the wide flange members. The HSS size  
493 was HSS16x0.625. The thickness of the flanges as well as the thickness of the web of the horizontal  
494 wide flange segments were selected to be the same as the modular connector. The length of the  
495 each segment was:  $l_s = e + c$ . These parameters of the modular connectors, the horizontal wide  
496 flange segments, and HSS could also be determined based on other structural constraints as well  
497 as using optimization approaches.

498 Figure 14 shows the buckling mode shape of the structure corresponding to the critical load

499 factor, that is the smallest eigenvalue of 5.1 determined from the FE analyses. As the load is mul-  
500 tiplied by this critical load factor, the structure experiences global buckling as shown in Figure 14.  
501 This mode shape is expected due to the parabolic, arch-type geometry of the grid shell. The con-  
502 siderably high critical load factor indicates a significant increase in the system buckling capacity in  
503 comparison to the results from the form-finding methodology from which the critical load factor is  
504 1. This difference can be attributed to the fact that TNA uses truss elements with no rotational stiff-  
505 ness and the modular connectors were not modeled. In contrast, in the FE analysis, the moment-  
506 resisting connection between the members and modular connectors is incorporated through the  
507 tie constraints. By explicitly modeling the modular connectors, the eccentricity of loading due to  
508 the starter segments is incorporated as well (eccentricity was not considered in the form-finding  
509 methodology). The high critical load factor indicates a satisfactory performance of the system  
510 even with the load eccentricity included. The FE analysis indicates that the novel modular con-  
511 nector provides an enhanced buckling resistance due to the ability to achieve a moment-resisting  
512 connection and demonstrates that the proposed form-finding methodology is conservative.

## 513 **CONCLUSION**

514 This paper proposed a new modular approach to address a major challenge in the design, fabri-  
515 cation and erection of steel grid shells: the nodal connector. More specifically, this research intro-  
516 duces a novel modular connector which can be easily prefabricated and readily available. A single  
517 modular connector can be used to achieve various grid shell forms. Thus, this paper proposed a  
518 new kit-of-parts approach comprised of: (1) prefabricated modular connectors, (2) standard wide  
519 flange sections, and (3) bolted splice connections that can be used for both free-form and form-  
520 found grid shells. This approach also provides a significant structural advantage as it achieves a  
521 moment-resisting connection allowing the load to be redistributed in case of a sudden member loss  
522 or replacement.

523 In this research, a new paradigm in design of steel grid shells was proposed: start with the  
524 nodal connector that is designed for ease of fabrication and then develop efficient structural forms  
525 that are consistent with the connector. Specifically, the development of free-form grid shells using

526 the kit-of-parts was first introduced. Additionally, a methodology for finding efficient grid shells  
527 that are compatible with the modular connector was proposed. By using a numerical form-finding  
528 technique that is consistent with the modular connector, coupled with structural and geometric  
529 constraints, the methodology develops a compression-only grid shells and selects the section size  
530 of the wide flange members while minimizing the self-weight of the structure. Linear eigenvalue  
531 buckling analyses using a high-fidelity FE numerical model were performed to investigate the  
532 global behavior of the developed structures. The results demonstrated the promise of the modular  
533 connector to achieve efficient forms while providing enhanced stability to the structure through the  
534 moment-resisting connection.

535 This research has focused on conceptually designing the modular connector, developing the  
536 form-finding methodology, and demonstrating the feasibility of the found forms through linear  
537 eigenvalue buckling analysis. Future research could focus on shape and sizing optimization of the  
538 modular connector. For example, a multi-objective optimization problem could be formulated to  
539 include material efficiency and fabrication/erection cost while meeting structural and transportation  
540 requirements. Additionally, prefabrication and construction strategies could be explored and the  
541 behavior of the system during erection should be investigated. A prototype could be built for better  
542 understanding the behavior of the system under construction and design loads. As the modular  
543 connector uses moment-resisting connections, there is a potential for the system to sustain higher  
544 loads, tolerate sudden member loss, and permit member replacement without compromising its  
545 structural integrity. Analysis to understand these behaviors would require a refinement of the  
546 the FE model to include material and geometric nonlinearities. A nonlinear material model can  
547 be incorporated, for example assuming an elastic-perfectly plastic stress-strain relationship or if  
548 needed, including strain hardening. The sudden member loss scenario can be simulated using  
549 different FE models, for example: (1) a member is severed, and dynamic analysis is performed  
550 that can capture the high velocity stress wave propagation through the wide flange member into the  
551 connector and (2) member is removed from the geometry and nonlinear static analysis is performed  
552 to investigate the behavior of the faulted structure. The second analysis could be used for evaluating

553 the structure for member replacement.

554 Ultimately, this research presents a new approach to the design of steel grid shell structures  
555 that “modularizes” the nodal connector between members allowing identical modular connectors  
556 to be used repeatedly throughout the structure and among many structures. This approach aims  
557 to reduce the time and cost for fabrication and erection, while also achieving a resilient structural  
558 design.

## 559 **DATA AVAILABILITY STATEMENT**

560 Some or all data, models, or code that support the findings of this study are available from the  
561 corresponding author upon reasonable request.

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**TABLE 1. Results of the case studies.**

$S$ (m)	$W_{min}$ (kN)	$h$ (m)	$z_c$ (m)	$F_{max}$ (kN)	$\mu$	Section size W14x	$\lambda_1$
45.7	1699	14.6	2.86	2078	0.253	233	1.01
53.3	2648	9.14	1.62	3049	0.246	257	1.02
61.0	3269	10.4	2.04	3500	0.317	257	1.05
76.2	4713	13.4	2.84	4762	0.505	257	1.02

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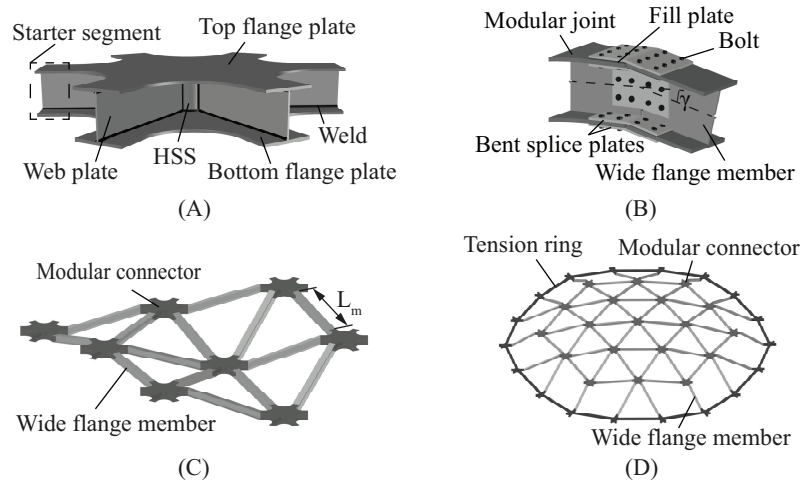
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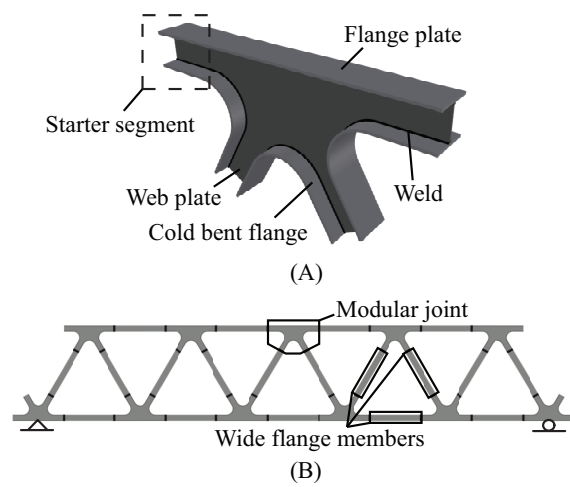
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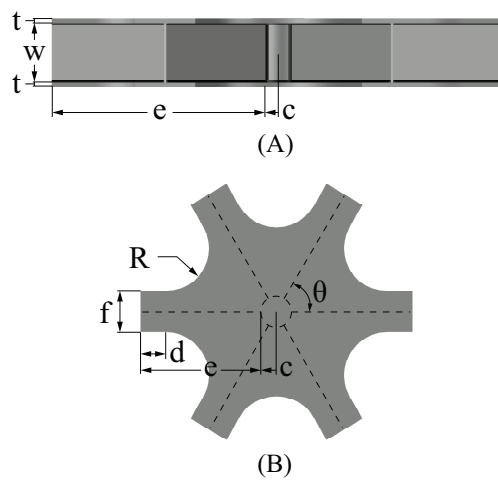


**FIG. 1. Modular approach to grid shells: (A) Modular connector, (B) Bolted splice connection, (C) Free-form grid shell, and (D) Form-found grid shell.**

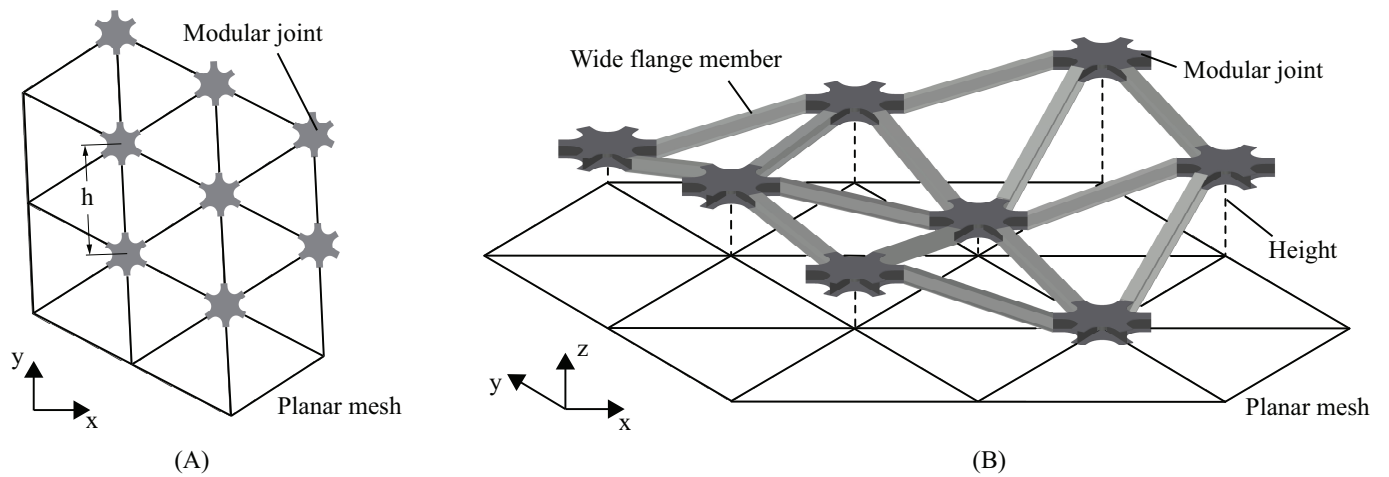




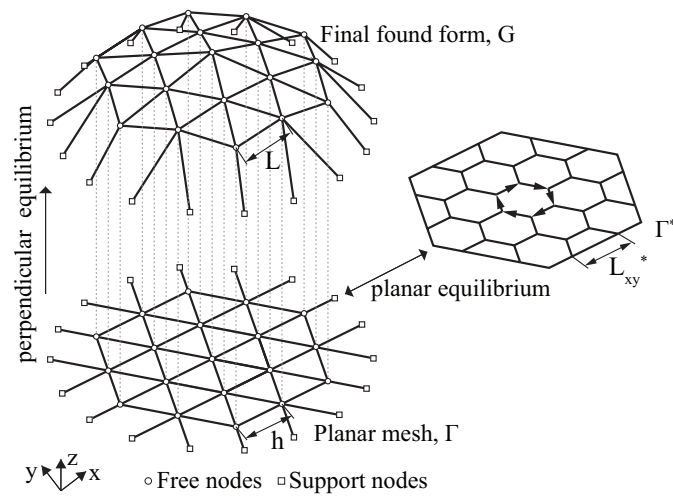
**FIG. 2. Modular approach to steel bridges: (A) Modular joint and (B) Constant-depth simply supported bridge with modular joints and standard wide flange members (adapted from Tumbava et al. (2021), ©ASCE).**



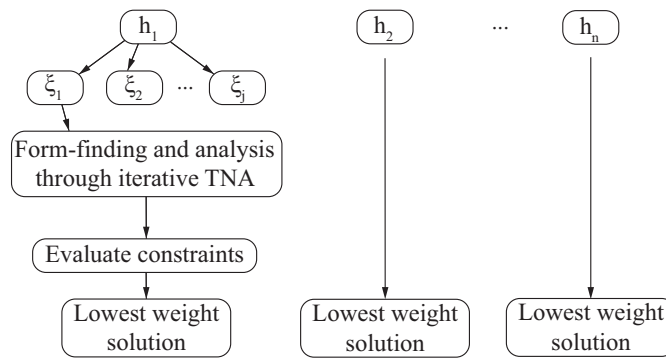
**FIG. 3. Geometric parameters of the modular connector: (A) Elevation view and (B) Plan view.**



**FIG. 4. Free-form grid shell: (A) Modular connectors projected onto a parallel planar mesh and (B) Isometric view of modular connectors connected to wide flange members (splice plates not shown for clarity) [(A) and (B) adapted from Tumbeva et al. (2018)].**



**FIG. 5. Form-found 3D equilibrium network,  $G$ , its planar projection the form diagram,  $\Gamma$ , and the reciprocal force diagram,  $\Gamma^*$ .**



**FIG. 6. Compression-only form-finding methodology overview.**

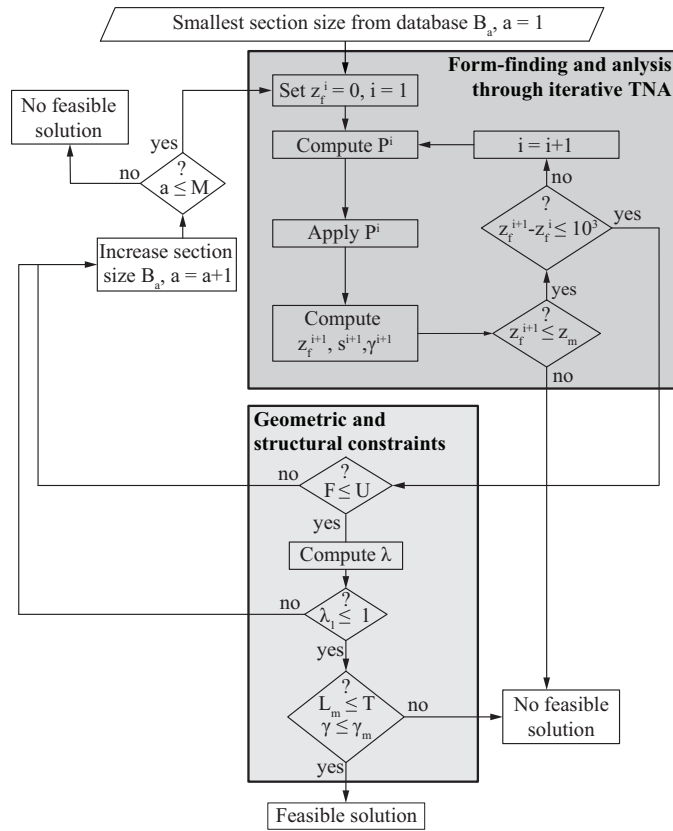
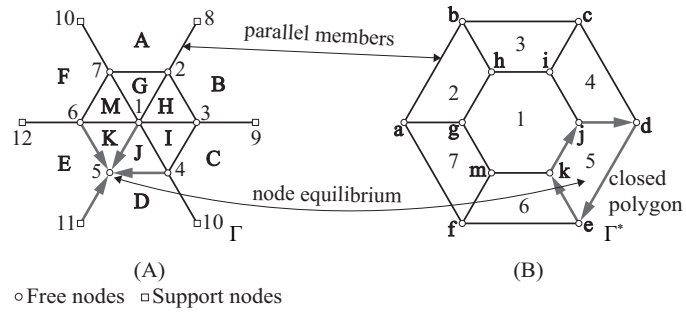
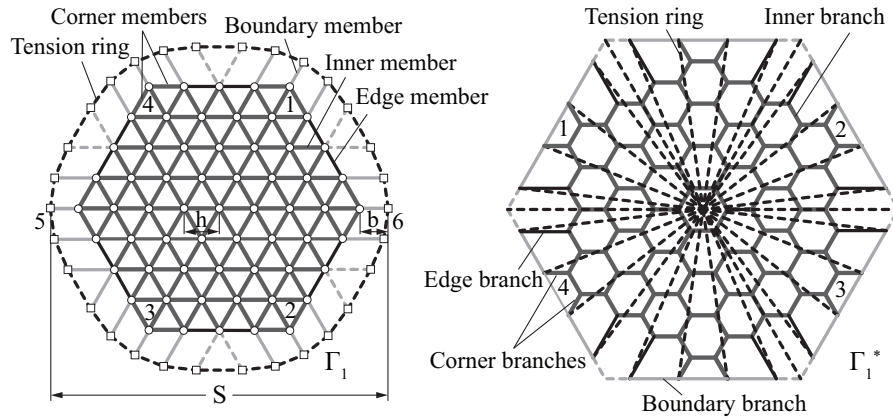


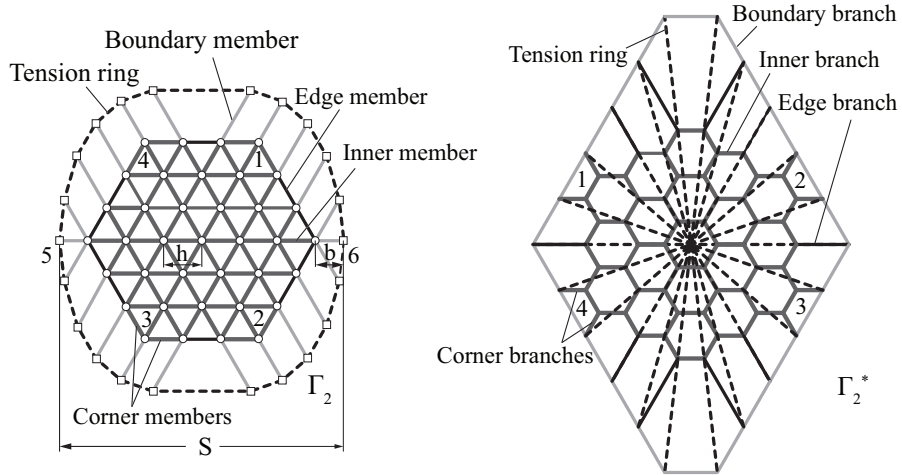
FIG. 7. Flow chart of the form-finding methodology for single combination of  $h$  and  $\xi$ .



**FIG. 8. Reciprocal diagrams: (A) Form diagram,  $\Gamma$  and (B) Force diagram,  $\Gamma^*$ .**



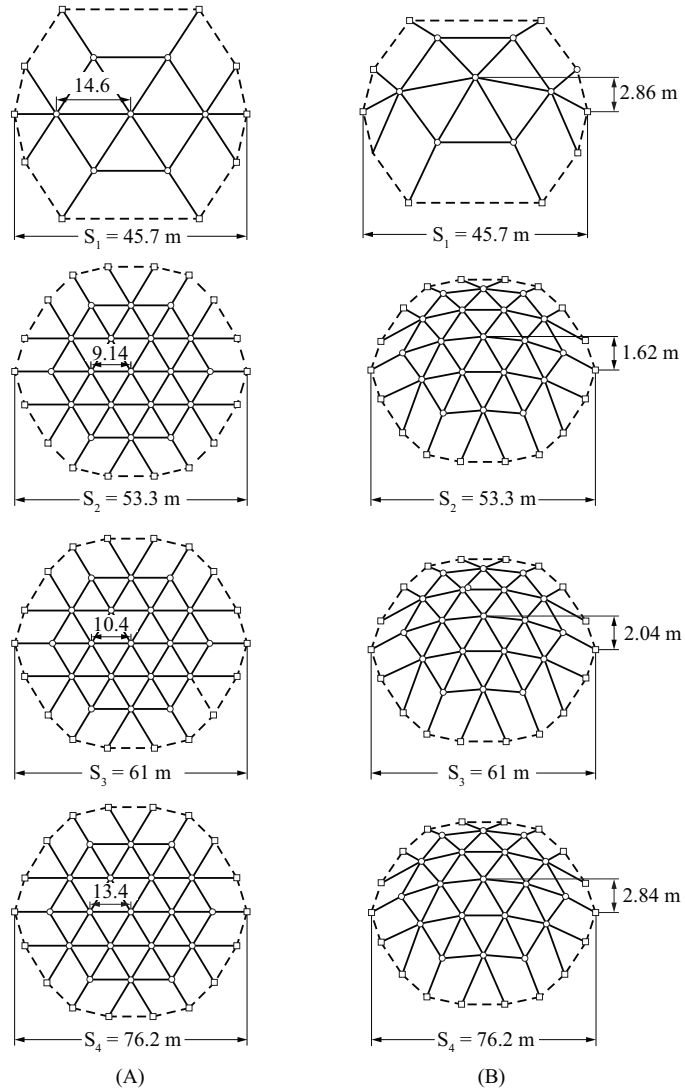
(A)



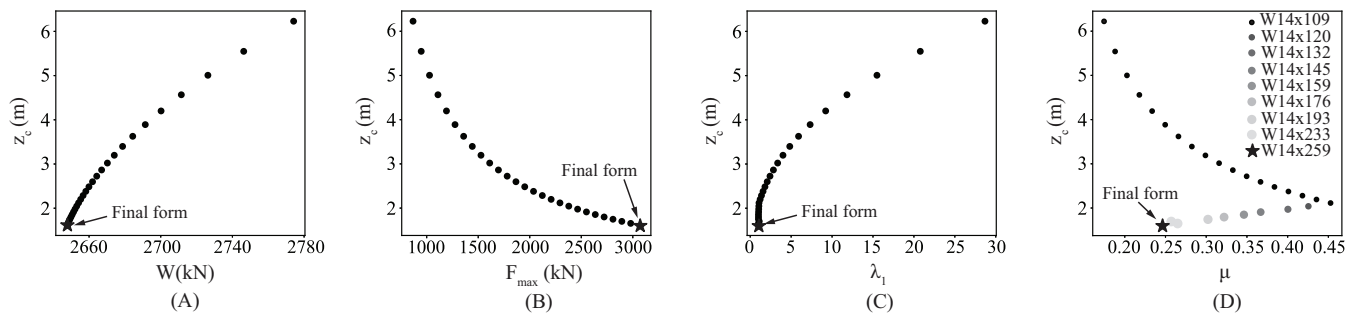
(B)

**FIG. 9. Development of form and force diagrams: (A) Type  $\Gamma_1, \Gamma_1^*$  and (B) Type  $\Gamma_2, \Gamma_2^*$ .**

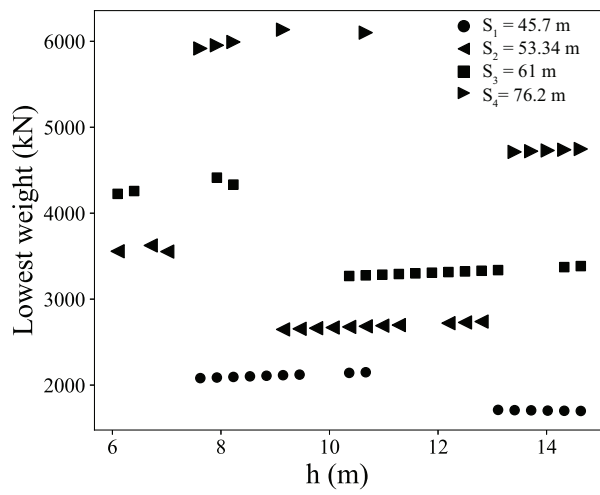




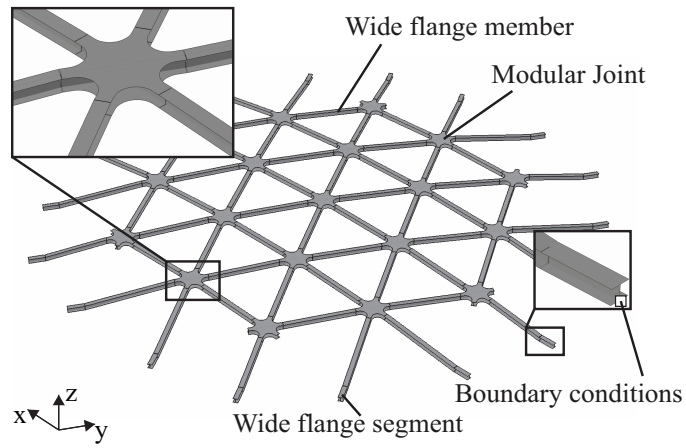
**FIG. 10. Results for each case study: (A) Planar mesh, the form diagram,  $\Gamma$  and (B) Form-found grid shell,  $G$  (not drawn to scale).**



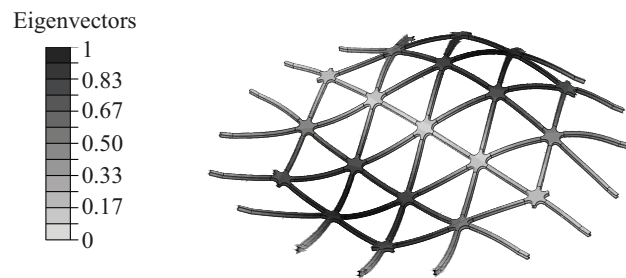
**FIG. 11. Results for  $S_2 = 53.3$  m and  $h = 9.14$  m based on: (A) Weight,  $W$ , (B) Largest axial force,  $F_{max}$ , (C) critical buckling load factor,  $\lambda_1$ , and (D) member buckling capacity factor,  $\mu$ .**



**FIG. 12. Lowest weight solution for each mesh size,  $h$ .**



**FIG. 13. Finite element model.**



**FIG. 14. Buckling mode shape corresponding to critical load factor of 5.1 determined from 3D FE linear eigenvalue analysis for span  $S_2 = 53.3$  m and  $h = 9.14$  m.**